



2020 Collegiate Design Series
SAE Aero Design Rules



Version 2020.0

Forward

Welcome to SAE Aero Design 2020! Our goal each year is to create and refine a set of competition events with relevant real-world design challenges that engineering students of all disciplines and levels of experience can enjoy. This competition provides the opportunity to apply your engineering knowledge in a team environment to create an original aircraft design. After the design phase, the team must build, test, and refine their aircraft while preparing to compete against other teams. Teams will validate their efforts by creating a design report and an oral presentation as well as demonstrating flight performance during competition.

SAE Aero Design consists of three competition classes: Regular class, Advanced class and Micro class. Regular class traces its lineage back to the very beginning of Aero Design, when it was created as a heavy lift competition for model aircraft. For 2020, Regular class has a new theme: teams are asked to develop a bush plane design that can operate from short runways while carrying outsized cargo as well as other cargo. The customer needs to transport many large spherical storage tanks to very remote locations, with short runways. Normal aircraft cannot perform this task and helicopters are too expensive and lack sufficient range. Additional constraints provided by the customer include: a 100 ft runway limit, a 120-inch wingspan limit, and a 1000-watt limiter.

Advanced class continues with the theme from 2018-2019 of colonizing a planet by delivering “colonists”, “supplies”, and “habitats”. “The “colonists” are delivered to a designated zone using a small autonomous glider, carried by the primary aircraft and released in a designated area. The primary aircraft also delivers the “habitats” and “supplies”. The previous year also saw the implementation of a 750-watt limiter. For 2020, the wingspan limit for the primary aircraft has been reduced to 132 inches.

Micro class continues with the same mission as created for 2017-2018, with some minor changes. The low-density industrial material for payload, the aircraft container, and mandatory timed assembly demo combine to give Micro class a demanding set of requirements.

The Rules Committee closely reviews the feedback teams submit after each Aero Design event and we make use of the survey when we consider and implement new rules and rule changes. Please read the 2020 rules closely. Also, please read the FAQs for 2020 that we will post in the question and answer forum. Lastly, please make use of the Aero Design question and answer forum to resolve questions about the rules.

The Rules Committee is excited about the 2020 events and we hope teams will enjoy this opportunity to apply their education to real world engineering tasks. We expect both 2020 events to sell out quickly, so be sure to apply as soon as possible. For this year, we were able to expand both Aero Design venues to accept 85 teams, giving 10 additional teams a chance to compete. If

you are waitlisted, please continue with your design effort, as we were able to accommodate nearly all waitlisted teams over the last few years as other teams dropped out.

*Everyone at SAE Aero Design wishes all teams the best of luck for Aero Design 2020!
SAE Aero Design Rules Committee*

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1 COMPETITION REQUIREMENTS

1.1 INTRODUCTION

Official Announcements and Competition Information

The SAE Aero Design competition is intended to provide undergraduate and graduate engineering students with a real-world design challenge. These rules were developed by industry professionals with a focus on educational value and hands-on experience. These rules were designed to compress a typical aircraft development program into one calendar year, following the early development phase of system engineering and requirements derivation. This competition will expose participants to the nuances of conceptual design, manufacturing, system integration/test, and verification through demonstration.

SAE Aero Design features three classes of competition—Regular, Advanced, and Micro.

1. **The Regular Class** is an all-electric class intended to develop a fundamental understanding of aircraft design.
2. **The Advanced Class** is an all-electric class designed to inspire future engineers to take a systems approach to problem solving, at the same time, exposing them to explore the possibilities of autonomous flights.
3. **The Micro Class** is an all-electric class designed to help students engage in trades between two potentially conflicting requirements, carrying the highest payload fraction possible, while simultaneously pursuing the lowest empty weight possible.

Other SAE Aero Design Competitions:

SAE BRASIL <http://www.saebrasil.org.br>

1.2 SAE AERO DESIGN RULES AND ORGANIZER AUTHORITY

General Authority

SAE International and the competition organizing bodies reserve the rights to revise the schedule of any competition and/or interpret or modify the competition rules at any time and in any manner, that is, in their sole judgment, required for the efficient and safe operation of the event or the SAE Aero Design series.

1. Penalties

SAE International and the competition organizing bodies reserve rights to modify the points and/or penalties listed in the various event descriptions; to accurately reflect the operations execution of the events, or any special conditions unique to the site.

2. Rules Authority

The SAE Aero Design Rules are the responsibility of the SAE Aero Design Rules Committee and are issued under the authority of the SAE International University Programs Committee. Official announcements from the SAE Aero Design Rules Committee, SAE International or the other SAE International Organizers shall be considered part of and have the same validity as these rules. Ambiguities or questions

concerning the meaning or intent of these rules will be resolved by the officials, SAE International Rules Committee or SAE International Staff.

3. Rules Validity

The SAE Aero Design Rules posted at www.saeerodesign.com/go/downloads and dated for the calendar year of the competition are the rules in effect for the competition. Rule sets dated for prior competition years are invalid.

4. Rules Compliance

By entering an SAE Aero Design competition, the team members, Faculty Advisors and other personnel of the entering university agree to comply with, and be bound by, the rules and all rules interpretations or procedures issued or announced by SAE International, the SAE Aero Design Rules Committee and other organizing bodies. All team members, Faculty Advisors and other university representatives are required to cooperate with and follow all instructions from Competition Organizers, officials and judges.

5. Understanding the Rules

Teams are responsible for reading and understanding the rules in their entirety for the competition in which they are participating. The section and paragraph headings in these rules are provided to facilitate reading: they do not affect the paragraph contents.

6. Loopholes

It is virtually impossible to anticipate a comprehensive design space that covers all possibilities and potential questions about the aircraft's design parameters or the conduct of the competition. Please keep in mind that safety remains paramount during any SAE International competition, so any perceived loopholes should be resolved in the direction of increased safety/concept of the competition. When in doubt, please contact the SAE Aero Design Rules Committee, using the FAQ forum, early to avoid design impacts at competition.

7. Participating in the Competition

Teams, team members as individuals, Faculty Advisors and other representatives of a registered university who are present on-site at a competition are considered to be "participating in the competition" from the time they arrive at the event site until they depart the site at the conclusion of the competition or earlier by withdrawing.

8. Visa--United States Visas

Teams requiring visas to enter to the United States are advised to apply at least sixty (60) days prior to the competition. Although most visa applications seem to go through without an unreasonable delay, occasionally teams have had difficulties and, in several instances, visas were not issued before the competition.

AFFILIATED CDS STUDENT TEAM MEMBERS WILL HAVE THE ABILITY TO PRINT OUT A REGISTRATION CONFIRMATION LETTER FOR THE INDIVIDUAL EVENT(S) THAT THEY ARE ATTENDING. ONCE A STUDENT TEAM MEMBER AFFILIATES THEMSELVES TO THEIR TEAM PROFILE PAGE UNDER THEIR INDIVIDUAL EDIT SECTION, THEY WILL HAVE THE OPPORTUNITY TO PRINT OUT THEIR

PERSONALIZED LETTER WITH THE FOLLOWING INFORMATION: STUDENT'S NAME, SCHOOL'S NAME, THE CDS EVENT NAME, OFFICIAL DATES AND LOCATION(S).

9. Letters of Invitation

Neither SAE International staff nor any Competition Organizers are permitted to give advice on visas, customs regulations or vehicle shipping regulations concerning the United States or any other country.

10. Certificates of Participation

SAE International and Competition Organizers do not create any Participation Certificates outside of the auto-generated certificate on your team profile page at sae.org.

Certificates are available as soon as students are affiliated to the current competition's team. Certificates will not be available once that competition year closes.

11. Violations of Intent

The violation of the intent of a rule will be considered a violation of the rule itself. Questions about the intent or meaning of a rule may be addressed to the SAE International Officials, Competition Organizers or SAE International Staff.

12. Right to Impound

SAE International and the other competition organizing bodies reserve the right to impound any on-site vehicle/aircraft at any time during a competition for inspection and examination by the Competition Organizers, officials and technical inspectors.

1.3 SOCIETY MEMBERSHIP AND ELIGIBILITY

1. Society Membership

Individual team members must be members of one of the following societies: (1) SAE International or an SAE International affiliate society, (2) ATA, (3) IMechE or (4) VDI. Proof of membership, such as a membership card, is required at the event. Students who are members of one of the societies listed above are not required to join any of the other societies in order to participate in any SAE competition. Students may join online at: <https://www.sae.org/participate/membership/join>

Teams are also required to read the articles posted on the SAE Aero Design News Feed (www.sae.org/participate/aerodesign/feed) published by SAE International and the other organizing bodies. Teams must also be familiar with all official announcements concerning the competition and rule interpretations released by the SAE Aero Design Rules Committee.

2. Team Pilots

Team pilots are not required to be students or SAE International members; however, all pilots must be current members of the Academy of Model Aeronautics or the Model Aircraft Association of Canada (AMA has an agreement with MAAC). Valid AMA membership cards must be presented at the flying field prior to flying any team's aircraft. Non-US pilots can obtain a discounted AMA Affiliate membership that

covers flying activities while in the US by going to the AMA web site and submitting the following form: <https://www.modelaircraft.org/files/902.pdf>.

1.4 LIABILITY WAIVER AND INSURANCE REQUIREMENTS

All on-site participants and Faculty Advisors are required to sign a liability waiver. Individual medical and accident insurance coverage is the sole responsibility of the participant.

1.5 RINGERS PROHIBITED

In order to maintain the integrity of the competition, the Faculty Advisor must prohibit ringers. A ringer is someone that has exceptional skills related to the competition (e.g., a professional model builder) that cannot be a legal member of the team but helps the team win points.

1.6 DESIGN AND FABRICATION

The aircraft must be designed and built by the SAE International student members without direct involvement from professional engineers, radio control model experts, pilots, machinists, or related professionals. The students may use any literature or knowledge related to R/C aircraft design and construction and information from professionals or from professors, as long as the information is given as discussion of alternatives with their pros and cons and is acknowledged in the references in the design report. Professionals may not make design decisions, nor contribute to the drawings, the report, or the construction of the aircraft. The Faculty Advisor must sign the Statement of Compliance given in the Appendix.

1.7 ORIGINAL DESIGN

Any aircraft presented for competition must be an original design whose configuration is conceived by the student team members. Photographic scaling of an existing model aircraft design is not allowed. Use of major components such as wings, fuselage, or empennage of existing model aircraft kits is prohibited. Use of standard model aircraft hardware such as motor mounts, control horns, and landing gear is allowed.

1.8 OFFICIAL LANGUAGES

The official language of the SAE Aero Design series is English. Document submissions, presentations and discussions in English are acceptable at all competitions in the series.

Team members, judges and officials at Non-U.S. competition events may use their respective national languages for document submissions, presentations and discussions if all the parties involved agree to the use of that language.

1.9 UNIQUE DESIGNS

Universities may enter more than one team in each SAE Aero Design competition, but each entry must be a unique design, significantly different from each other. If the aircraft are not significantly different in the opinion of the Rules Committee and Organizer, then the university will be considered to have only a single entry and only one of the teams and its aircraft will be allowed to participate in the competition. For example, two aircraft with identical wings and fuselages but different empennage would likely not be considered

significantly different. For guidance regarding this topic, please submit a rules question at www.saeerodesign.com.

1.10 AIRCRAFT CLASSIFICATION/DUPLICATE AIRCRAFT

1. One Team Entry per Class

A university is limited to registering one team per class.

2. Backup Aircraft

When a team has an identical aircraft as a back-up, the back-up aircraft must go through inspection with the primary aircraft.

1.11 AIRFRAME ELIGIBILITY

Airframes will only be allowed to compete during a single academic year. An airframe may be entered in both SAE Aero Design East and SAE Aero Design West during the same calendar year, but that same airframe may not be used in either competition during the following year. Entering the same airframe in SAE Aero Design West one year and SAE Aero Design East the next year is not allowed.

An airframe is considered entered to competition during an academic year, once documentation on the design is submitted. If the airframe does not fly at competition during that same academic year, the airframe is not eligible for competition during future academic years.

The airframe must have been designed within eleven (11) months of competition and constructed within nine (9) months of competition. The airframe is defined as the fuselage, wings, and tail.

1.12 REGISTRATION INFORMATION, DEADLINES AND WAITLIST (NEW)

Teams intending to participate in the 2020 SAE Aero Design competitions must register their teams online per the open registration schedule shown in Table 1.1.

Table 1.1 Open Registration Schedule

Event	Start (Open)	End (Closed)
<i>Pre-Registration Window</i>	October 7, 2019 08:00 AM EDT	October 14, 2019 09:00 AM EDT
<i>Registration Window 1</i>	October 14, 2019 10:00 AM EDT	October 27, 2019 11:59 PM EDT
<i>Registration Window 2</i>	October 28, 2019 10:00 AM EDT	November 27, 2019 11:59 PM EDT

The registration fee is non-refundable and failure to meet these deadlines will be considered a failure to qualify for the competition. Separate entry fees are required for the East and West events.

1. **Team/Class/University Policy**

A university or college can only have one aircraft registered for one class. A university cannot register more than one team per class. The registration fees indicated on the website (\$1100) must be paid within 48 hours of registration.

2. **Individual Registration Requirements – ACTION REQUIRED**

All participating team members and Faculty Advisors must be sure that they are individually affiliated to their respective school / university on the SAE International website (www.sae.org) Team Profile page.

If you are not an SAE International member, go to www.sae.org and select the “Membership” link. Students will need to select the “Student Membership” link and then follow the series of questions that are asked. Please note: all student participants must be members of one of the organizations listed in Section 1.3.1 to participate in the events.

Faculty members who wish to become SAE International members should choose the “Professional Membership” link. Please note: this is not mandatory for Faculty Advisors.

All student participants and Faculty Advisors must affiliate themselves to the appropriate team(s) online.

The “Add New Member” button will allow individuals to access this page and include the necessary credentials. If the individual is already affiliated to the team, simply select the Edit button next to the name. Please be sure this is done separately for each of the events your team has entered.

All students, both domestic and international, must affiliate themselves online prior to the competition.

New for 2020: Each team member may participate for only one team. If the university or college is entering multiple classes, team members must choose only one team to affiliate with and participate in the competition with. For example, students cannot compete as part of a Micro class team and an Advanced class team.

3. **Pre-Registration Information**

The Pre-Registration Window is available to any university that placed Top 3 in any class during the previous competition season or won a Pre-Registration Pass at the previous competition. Pre-registration allows a school to register for multiple events but only in the class in which they qualified for Pre-Registration (Top 3 placement).

In Registration Window 2, any university can register at any competition in any class they have not already registered for without restriction as long as spots remain available. If a university earns one or more Pre-Registration pass and wishes to NOT take advantage of the opportunity, they can forfeit the pass and enter Registration Window 1 with no restrictions.

****NOTE: When your team is registering for a competition, only the student or Faculty Advisor completing the registration needs to be linked to the school. All other students and faculty can affiliate themselves after registration has been completed;**

however, this must be completed no later than two weeks before the competition start date.

1.13 WAITLIST

Once an event reaches the venue's capacity, all remaining registered team(s) will be asked to be placed on a waitlist. The waitlist is capped at 50 available spaces per event and will close on the same day as registration closes. Once a team withdraws from an event, an SAE International Staff member will inform your team by email (the individual who registered the team to the waitlist) that a spot on the registered teams list has opened. You will have 24 hours to accept or reject the position and an additional 24 hours to have the registration payment completed or process for payment begun. Waitlisted teams are required to submit all documents by the deadlines in order to be considered serious participants and any team that does not submit all documents will be removed from the waitlist.

1.14 POLICY DEADLINE

1. Failure to meet deadlines

Teams registering for SAE Aero Design competitions are required to submit several documents prior to the competition including a Design Report and Technical Data Sheet that the event judges use to evaluate the team during the competition. When these documents are not submitted, judges cannot properly assess the team. Additionally, teams that do not submit required documents typically do not come to the competition. Teams that do not notify us that they are withdrawing create the following problems:

- They are included in the static event schedules and judging time is wasted.
- Their unused registration slot cannot be offered to a team on the waitlist. Additionally, failure to submit the required documents is a clear violation of the rules.

2. Late Submission Penalty

Late submission or failure to submit the Design Report by the deadline will be penalized five (5) points per day. If your required documents are received more than five (5) days late, the documents will be classified as "Not Submitted" and your team will not be allowed to participate. Additionally, the automatic withdrawal policy will be in effect.

3. Automatic Withdrawal Policy

Failure to submit the required Design Report, Technical Data Sheet, and Drawings within five (5) days of the deadline will constitute an automatic withdrawal of your team. Your team will be notified before or on the 4th day of no submission that we have not received your documents and after the 5th day your team's registration will be canceled, and no refund will be given.

1.15 FACULTY ADVISOR

Each team is expected to have a Faculty Advisor appointed by the university. The Faculty Advisor is expected to accompany the team to the competition and will be considered by competition officials to be the official university representative. Faculty Advisors may advise

their teams on general engineering and engineering project management theory but may not design any part of the vehicle nor directly participate in the development of any documentation or presentation. Additionally, Faculty Advisors may neither fabricate nor assemble any components nor assist in the preparation, maintenance, or testing of the vehicle. In brief, Faculty Advisors may not design, build or repair any part of the aircraft. Faculty Advisors that are not eligible student team members may not participate in flight operations during competition weekend except as noted.

1.16 QUESTIONS, COMPLAINTS AND APPEALS

1. Questions

Any questions or comments about the rules should be brought to the attention of the Rules Committee by submitting a rules question at <https://www.saeerodesign.com>.

General information about hotels and other attractions in the area, as well as a schedule of events, will be posted on the SAE International website according to the competition in which you are competing: <https://www.sae.org/attend/student-events/>

2. Complaints

Competition officials will be available to listen to complaints regarding errors in scoring, interpretation, or application of the rules during the competition. Competition officials will not be available to listen to complaints regarding the nature, validity, or efficacy of the rules themselves at the competition. In other words, the Organizer will not change the rulebook at the field, unless the safety of the competition requires updates.

3. Appeal / Preliminary Review

A team can only appeal issues related to scoring, judging, venue policies, and/or any official actions *regarding their own team*. Team Captain(s) and/or Faculty Advisor must bring the issue to the Organizer's or SAE International staff's attention for an informal preliminary review before filing an official appeal.

A team cannot file an appeal to cause harm to another team's standing and/or score.

4. Cause for Appeal

A team may appeal any rule interpretation, own-team scoring or official actions which the team feel has caused some actual, non-trivial, harm to own-team, or has had a substantive effect on their score.

Teams may not appeal rule interpretations or actions that have not caused the team any substantive damage.

5. Appeal Format

If a Faculty Advisor or Team Captain(s) feel that their issue regarding an official action or rules interpretation was not properly addressed by the **event officials**, the team may file a formal appeal to the action or rules interpretation with the Appeals Committee.

All appeals must be filed in writing (see Appendix E) to the Organizer by the Faculty Advisor or Team Captain(s) only.

All appeals will require the team to post twenty-five (25) points as collateral. If the appeal is successful and the action is reversed, the team ***will not*** forfeit the twenty-five (25) collateral points. If the appeal is overruled, the team will forfeit the twenty-five (25) collateral points.

All rulings issued by the Appeals Committee are final.

6. Appeals Period

All appeals must be submitted within thirty (30) minutes of the end of the flight round or other competition event to which the appeal relates.

7. Appeals Committee

When a timely appeal is received, the committee will review the claims. All contentions or issues raised in the formal appeal will be addressed in a timely manner. The consideration in each review is whether the actions in dispute were just and in-line with the intent of the rules. Once the review is completed, a new order will be issued affirming, reversing or modifying the original determination.

All rulings issued by the Appeals Committee are final.

The Appeals Committee must consist of a minimum of three members: the Organizer or delegate, SAE International representative, and either the Chief Steward, the Chief Judge, the Air Boss and/or Rules Committee member.

1.17 PROFESSIONAL CONDUCT

1. Unsportsmanlike Conduct

In the event of unsportsmanlike conduct by team members or a team's Faculty Advisor, the team will receive a warning from a Competition Official. A second violation will result in expulsion of the team from the competition and loss of any points earned in all aspects of the competition.

2. Arguments with Officials

Arguments with or disobedience toward any competition official may result in the team being eliminated from the competition. All members of the team may be immediately escorted from the grounds.

3. Alcohol and Illegal Material

Alcoholic beverages, illegal drugs, firearms, weapons, or illegal material of any type are not permitted on the event sites at any time during the competition. Any violations of this rule will result in the immediate expulsion of all members of the offending school, not just the team member(s) in violation. This rule applies to team members and Faculty Advisors. Any use of illegal drugs or any use of alcohol by an underage person must be reported to the local law enforcement authorities for prosecution.

4. Organizer's Authority

The Organizer reserves the exclusive right to revise the schedule of the competition and/or to interpret the competition rules at any time and in any manner required for efficient operation or safety of the competition.

5. Ground Safety and Flight Line Safety Equipment

- **No open toe shoes allowed.** All team participants, including Faculty Advisors and pilots, will be required to wear CLOSED toe shoes during flight testing and during flight competition.
- **Smoking is prohibited.** Smoking is prohibited in all competition areas.
- All students in all classes involved at the flight line must wear safety glasses.
- Micro Class must wear hard hats in addition to safety glasses at the flight line.

1.18 SAE TECHNICAL STANDARDS ACCESS

A cooperative program of SAE International's Education Board and Technical Standards Board is making some of SAE International's Technical Standards available to teams registered for any North American CDS competition at no cost. The Technical Standards referenced in the Collegiate Design Series rules, along with other standards with reference value, will be accessible online to registered teams, team members and Faculty Advisors.

To access, teams can follow these procedures. Once registered, a link to SAE MOBILUS will appear to access the technical standards under "Design Standards" on your team's profile page on sae.org, where all the required onsite team information is added. On SAE MOBILUS, you will have the ability to search standards either by J-number assigned or topic of interest such as brake light.

A list of accessible SAE International Technical Standards can be found in Appendix F.

2 GENERAL AIRCRAFT REQUIREMENTS

2.1 AIRCRAFT IDENTIFICATION

Team number as assigned by SAE International must be visible on both the top and bottom of the wing, and on both sides of the vertical stabilizer or other vertical surface.

1. Aircraft must be identified with the school name, mailing address, and email address either on the outside or the inside of the aircraft.
2. Team numbers on Regular aircraft shall be a minimum of 3 inches in height.
3. Team numbers on the Advanced Class primary aircraft shall be a minimum of 3 inches in height. Team numbers on the Advanced Class Colonist Delivery Aircraft (CDA) shall be a minimum of 1 inch in height.
4. Team numbers on Micro Class shall be a minimum of 1 inch in height.
5. The University name must be clearly displayed on the wings or fuselage.
6. The University initials may be substituted in lieu of the University name provided the initials are unique and recognizable.

The assigned aircraft numbers appear next to the school name on the “Registered Teams” page of the SAE Aero Design section of the Collegiate Design Series website at:

SAE Aero East: <https://www.sae.org/attend/student-events/sae-aero-design-east>

SAE Aero West: <https://www.sae.org/attend/student-events/sae-aero-design-west>

2.2 PROHIBITED AIRCRAFT CONFIGURATION

Competing designs are limited to fixed wing aircraft only. Lighter-than-air aircraft, rotary wing aircraft such as helicopters or auto-gyros and steerable parafoil aircraft are not allowed to compete.

2.3 EMPTY CG DESIGN REQUIREMENT AND EMPTY CG MARKINGS ON AIRCRAFT

All aircraft must meet the following Center of Gravity (CG) related requirements:

1. All aircraft must be flyable at their designated Empty CG position (no payload, ready to fly) on the submitted 2D aircraft drawing.
2. All aircraft must have the fuselage clearly marked on both sides with a classic CG symbol (Figure 2.1) that is a minimum of 0.5 inches in diameter centered at the Empty CG position, per the submitted 2D drawings. (Wing type aircraft may place the two CG markings on the bottom of the wing.)
3. The Empty CG location will be verified during Technical and Safety Inspection.
4. No empty weight flight is required.



Figure 2.1 – Center of Gravity Symbol

2.4 GROSS WEIGHT LIMIT

Aircraft gross take-off weight may not exceed fifty-five (55) pounds.

2.5 CONTROLLABILITY

- All aircraft must be controllable in flight.
- If an aircraft is equipped with a wheeled landing gear, the aircraft must have some form of ground steering mechanism for positive directional control during takeoffs and landings. Aircraft may not rely solely on aerodynamic control surfaces for ground steering.

2.6 RADIO CONTROL SYSTEM

The use of a 2.4 GHz radio control system is required for all aircraft. The 2.4 GHz radio control system must have a functional fail-safe system that will reduce the throttle to zero **immediately** if the radio signal is lost. Teams may have to reset the default on the fail-safe to meet this requirement.

2.7 SPINNERS OR SAFETY NUTS REQUIRED

All powered aircraft must utilize either a spinner or a rounded model aircraft type safety nut.

2.8 METAL PROPELLERS

Metal propellers are not allowed.

2.9 LEAD IS PROHIBITED

The use of lead in any portion of aircraft (payload included) is strictly prohibited.

2.10 PAYLOAD DISTRIBUTION

The payload cannot contribute to the structural integrity of the airframe, meaning, the airframe must be able to fly without the payload installed.

2.11 STATIC PAYLOAD ATTACHMENT

All static payload must be secured with metal hardware that penetrates all payload plates. Payload plates must also be secured to the aircraft structure with metal hardware as a single mass inside the designated payload bay, as defined by each class.

2.12 AIRCRAFT BALLAST

Aircraft ballast is allowed. Ballast cannot be in the payload bay and must be properly secured.

2.13 STORED ENERGY RESTRICTION

Aircraft must be powered by the motor on board the aircraft. No other internal and/or external forms of stored potential energy allowed to include rubber bands and pressure vessels like CO2 cartridges.

2.14 CONTROL SURFACE SLOP

Aircraft control surfaces and linkages must not feature excessive slop. Sloppy control surfaces lead to reduced controllability in mild cases, or control surface flutter in severe cases.

2.15 SERVO SIZING

Analysis and/or testing must be described in the Design Report that demonstrates the servos are adequately sized to handle the expected aerodynamic loads during flight.

2.16 CLEVIS KEEPERS

All control clevises must have additional mechanical keepers to prevent accidental opening of the control clevis in flight.

2.17 RED ARMING PLUG

All electric powered aircraft MUST use a discrete and removable red arming plug to arm and disarm the aircraft propulsion system. This red arming plug must be integrated into the electrical circuit between the battery and the electronic speed controller (ESC).

1. The red arming plug must physically be located at 40% to 60% of the aircraft length from the aircraft propeller. This is to allow arming and disarming the aircraft at a safe distance from the propeller.
2. The red arming plug must be located on top of the fuselage or wing and external of the aircraft surface.
3. The location of the red arming plug must be clearly visible.
4. The non-removable portion of the arming plug interface may not have more than one male lead.
5. Disconnecting wiring harnesses to arm and disarm a system will NOT be allowed.

2.18 REPAIRS, ALTERATIONS, AND SPARES

1. The original design of the aircraft as presented in the written and oral reports must be maintained as the baseline aircraft during the competition.
2. In the event of damage to the aircraft, the aircraft may be repaired provided such repairs do not drastically deviate from the original baseline design. All major repairs must undergo safety inspection before the aircraft is cleared for flight.

2.19 ALTERATION AFTER FIRST FLIGHT

Minor alterations are allowed after the first and subsequent flight attempts.

1. A penalty will be assessed ONLY if 2/3 of the ruling committee (Event Organizer, Head scoring judge and/or SAE staff judge) agree that there were significant modifications made from the baseline configuration.
2. If the ruling committee determines that the changes are a result of safety-of-flight, the changes will not incur penalty points. Alteration must be reported utilizing Engineering Change Request (ECR) Appendix C.

3 MISSION REQUIREMENTS AND SCORING

3.1 AIR BOSS

The Air Boss is a qualified SAE event official or appointed volunteer that manages the flight line process. Their responsibilities include:

- Ensure the safety of the flight line through maintaining an orderly and controlled runway.
- Be the official of record for the success or failure of the aircraft's flight, including takeoff and landing.
- Declare termination of flight at any time during the attempt.
- Air Boss, or event organizers, may continue flight operations at their discretion in continuous winds up to 45 knots with gusts no greater than 65 knots.

3.2 PILOT STATION(S)

Pilot area will be defined at pre-competition meeting (Friday Night All-hands). All pilots must fly from designated area.

3.3 ROUND ATTEMPT

Teams are allowed one (1) flight attempt per round. There is no fixed or guaranteed number of flight rounds. The number of rounds will depend on local conditions.

- **Regular and Advanced Classes:** Without violating other take-off restrictions, a team can have multiple attempts to become airborne within the team's prescribed time limit for each respective class identified in Section 3.8.
- **Micro Class:** only one launch attempt is allowed per round.

3.4 MOTOR RUN-UP BEFORE TAKE-OFF

Aircraft may be throttled up/run up for take-off, subject to the following conditions:

- **Advanced Class:** Use of one (1) helper to hold the aircraft for take-off is allowed. The helper may not push the aircraft on release.
- **Regular Class:** Use of one (1) helper to hold the aircraft is allowed. Main wheels must be placed on the take-off line for Regular Class. The helper may not push the aircraft upon release.
- **Micro Class:** Aircraft must be run up and hand launched within the launch area for Micro.

3.5 AIRCRAFT CONFIGURATION AT LIFTOFF AND DURING THE FLIGHT ATTEMPT

The aircraft must remain intact during a flight attempt to receive full flight score. A flight attempt includes activities at the starting line, the take-off roll, take-off, flight, landing and recovery after landing.

A twenty-five percent (25%) deduction from the flight score will be assessed if any of the following items are observed to completely detach from the aircraft during a flight attempt.

- Stickers

- Tape
- Coverings

Except for a broken prop during landing, if any other components fall off the during a flight attempt, the flight will be disqualified.

3.6 COMPETITION CIRCUIT REQUIREMENTS

1. During departure and approach to landing, the pilot must not fly the aircraft in a pattern that will allow the aircraft to enter any of the no-fly zones.
2. No aerobatic maneuvers will be allowed at any time during the flight competition in any competition class. This includes but not limited to: loops, figure 8's, Immelmann, all types of rolling maneuvers and inverted flight.
3. Regular and Micro Class aircraft must successfully complete a minimum of one 360° circuit. See Table 3.2 for additional information.
4. Advanced Class has no specific flight pattern. (See Advanced Class rules for details concerning the releasable payload drop mission element.)

3.7 BATTERY PROTECTION

- All batteries in the aircraft must be positively secured so that they cannot move under normal flight loads.
- The battery bay or location in the aircraft must be free of any hardware or other protrusions that could penetrate the battery in the event of a crash.

3.8 TIME LIMITS AND MULTIPLE FLIGHTS ATTEMPTS

- Multiple takeoff attempts are allowed within the class-specific time allotment as long as the aircraft has NOT become airborne during an aborted attempt. Refer to Table 3.1 for additional information regarding multiple takeoff attempts.
- If an airborne aircraft returns to the ground after being airborne and is beyond the take-off limits, the flight attempt will be disqualified for that round.

Table 3.1: Flight Attempt Information

Class	Time Limit (sec)	Can make multiple take-off attempts if:			Take-off Attempt is defined as the point at which:
		Still within the Time Limit	Bounce within required take-off distance	Bounce outside the required take-off distance	
Regular	120	Yes	Yes	No	The main wheels leave the starting line
Advanced	180	Yes	Yes	No	The aircraft moves forward under its own power
Micro	60	No	No	No	The launcher is no longer in contact with the aircraft

3.9 TAKE-OFF

Take-off direction will be determined at the discretion of the Air Boss. If possible, the take-off direction will face into the wind. Changes in wind direction, in light and variable winds, may affect the take-off direction during the middle of a competition class flight round. SAE Aero Design reserves the right to change the take-off direction at any time during a flight round for weather or safety reasons.

1. Regular and Advanced Class aircraft must remain on the runway during the take-off roll.
2. Micro Class must be hand launched from the designated launch area.
3. Distance requirements are defined in Table 3.2.
4. Making the initial turn before passing the “distance from initial start before turn” requirement will disqualify that flight attempt.

Table 3.2: Take-off Information

Class	Take-Off Distance Limits (ft.)	Distance from initial start before turn (ft.)	Description
Regular	100 ft.	400 ft.	Aircraft must be airborne within the prescribed take-off distance.
Advanced	None	None	Aircraft will have the full use of the runway.
Micro	N/A	400 ft.	Team may use the entire launch area per attempt to get the aircraft airborne. Only one (1) launch attempt per round is allowed.

3.10 LANDING REQUIREMENTS

A successful landing is defined as a controlled return to the ground. Aircraft must remain inside the specified landing zone for each class. The airplane may leave the landing zone only if given permission by the Air Boss.

The landing zone is a pre-determined fixed area for each class for the purpose of returning a flying aircraft to the ground. See Table 3.3 for class requirements.

1. The landing zones will be visibly marked at the site prior to the start of competition.
2. It is the team and team pilot's responsibility to be aware of the class-specific landing zone dimensions at the event site.
3. Any aircraft that leaves their designated landing zone or the paved runway for any reason during landing are subject to a penalty of fifty percent (50%) of any points earned during the flight prior to landing.
4. Any flight where the aircraft does not make the initial touch down for landing inside the designated landing zone is disqualified.
5. Touch-and-go landings are not allowed and will be judged as a failed landing.
6. The criterion for being within the landing zone is that no supporting part of the aircraft that is touching the ground can be outside the landing zone. For example, a wing tip or fuselage can overhang the edge of the landing zone, as long as no supporting part of the aircraft is physically touching outside the landing zone.

Table 3.3: Landing Distance Limit

Class	Landing Distance Limits (ft.)	Description
Regular	400 ft.	Aircraft must land in the same direction as take-off and stop within the designated landing zone.
Advanced	Available Runway	Aircraft must land in the same direction as take-off and stop within the designated landing zone.
Micro	200 ft.	Aircraft must land in the same direction as take-off and stop within the designated landing zone.

3.11 GROUNDING AN AIRCRAFT

1. An aircraft will be grounded if it is deemed non-flight-worthy or not in compliance with class rules by any SAE official, event official or a designated technical/safety inspector.
2. Until the non-flight-worthy or out of compliance condition has been addressed and has been cleared by re-inspection, the aircraft will not be allowed to fly in the competition.

3.12 NO-FLY ZONE

Each competition will have venue-specific **no-fly zones**. The no-fly zones will be defined during the all hands briefing at the event and during the pilot's briefings.

1. At no time will an aircraft enter the no-fly zones, whether under controlled flight or uncontrolled.
2. The first infraction for crossing into the no-fly zone will result in an invalidated flight attempt and zero points will be awarded for that flight.
3. A second infraction will result in disqualification from the entire event and loss of all points.
4. It is the team and team pilot's responsibility to be aware of the venue-specific no-fly zones and to comply with all venue specific rules.
5. If a team is unable to directionally control their aircraft and it is headed towards or is in a no-fly zone, the Judges and/or Air Boss may order the pilot to intentionally crash the aircraft to prevent it from endangering people or property. This safety directive must be followed immediately, if ordered by the officials.

3.13 FLIGHT RULES ANNOUNCEMENT

Flight rules will be explained before the flight competition begins, either during the pilots' meeting or during activities surrounding the technical inspections and oral presentations.

3.14 FLIGHT RULES VIOLATIONS

1. Violation of any flight rule may result in the team being eliminated from the competition.
2. All members of an eliminated team may be escorted from the grounds.

3.15 LOCAL FIELD RULES

In addition to competition rules, the local flying club may have additional rules in place at the event flying field.

1. Club rules will be obeyed during the flight competition.
2. If club rules conflict with competition rules, it is the responsibility of the Team Captain(s) and/or Faculty Advisor to bring attention to the conflict and follow the appeals process to resolve the conflict.

3.16 COMPETITION SCORING

A team's final, overall score is composed of scores in the following categories:

1. Technical Design Report (Design, Written and Drawing)
2. Presentation
3. Flight Score
4. Penalties

Any Penalty Points assessed during the competition will be deducted from a team's overall score.

4 DESIGN REPORT

The Design Report is the primary means in which a team conveys the story of how their aircraft is the most suited design to accomplish the intended mission. The Design Report should explain the team's thought processes and engineering philosophy that drove them to their conclusions.

Some topics that are important to cover are: selection of the overall vehicle configuration, wing planform design including airfoil selection, drag analysis including three-dimensional drag effects, aircraft stability and control, power plant performance including both static and dynamic thrust, and performance prediction. Other topics should be included as appropriate. See the SAE Aero Design Report Guidelines available at www.saeerodesign.com/go/downloads for additional comments, suggested topics, and a suggested outline. For more information regarding performance prediction, a white paper by Leland Nicolai is also available at <http://www.saeerodesign.com/go/downloads>

4.1 SUBMISSION DEADLINES

The Technical Design Report, 2D drawing, and supplemental Tech Data Sheet (TDS) must be electronically submitted to www.saeerodesign.com no later than the date indicated on the Action Deadlines given on the SAE International Website:

<https://www.sae.org/attend/student-events>

Neither the Organizer nor the SAE International is responsible for any lost or misdirected reports, drawings, or server routing delays. The SAE International will not receive any paper copies of the reports through regular mail or email outside of the emergency submissions email.

4.2 ORIGINAL WORK

The Technical Design Report shall be the team's original work for the current competition year. Resubmissions of **previous and current** year's design reports will not be accepted. Recitation of previous year's work is acceptable **if and only if** appropriately cited and credited to the original author(s). Plagiarism is a forbidden industry and academic practice. All references, quoted text, and reused images from any source shall have appropriate citation within the text and within the Technical Design Report's Table of References, providing credit to the original author and editor.

Reports may be checked against **previous and current** years submissions to determine if re-use, copying, or other elements of plagiarism are indicated.

For the purposes of the SAE International Aero Design Competition, plagiarism is defined as any of the following:

- 1 Use of information from textbooks, reports, or other published material without proper citation
- 2 Use of sections or work from previous SAE Aero Design competitions without proper citation

If plagiarism is detected in the written report, a team will be given 24 hours to make a case to SAE and the SAE Aero Design Rules Committee. If the report and/or case is found to be insufficient, the team will receive zero score for the report. The team will be allowed to compete in all remaining categories of the competition but will not be eligible for awards. SAE also reserves the right to notify the University of the situation.

If plagiarism is detected in the oral presentation, team will receive zero score for the presentation. The team will be allowed to compete in all remaining categories of the competition but will not be eligible for awards. SAE also reserves the right to notify the University of the situation.

The SAE Aero Design Rules Committee & SAE International has the sole discretion to determine whether plagiarism is indicated, and the above rules are enacted. The above rules may be implemented at any time before, during, or for up to six (6) months after the competition event.

4.3 TECHNICAL DESIGN REPORT REQUIREMENTS

Technical Design Report will be 50 points (pts) of the competition score as broken down in Table 4.3.1.

- The Technical Design Report shall not exceed thirty (30) pages, including the certificate of compliance, 2D Drawing, and the Supplemental Datasheet for each class. If the design report exceeds thirty (30) pages, the judges will only score the first thirty (30) pages.
- The Technical Design Report shall include a Cover Page with Team Name, Team Number, and School Name and Team Member Names.
- The Technical Design Report shall include a Certificate of Compliance signed by hand by the team's Faculty Advisor.
- The Technical Design Report shall be typewritten and double-spaced. Tables, charts and graphs are exempt from this. For single-spaced reports, only the first fifteen (15) pages will be scored by judges. All other content sections will receive a zero (0).
- The report font shall be 12 pt. proportional; or 10 char/in. non-proportional font.
- The report margins shall be: 1" Left, 0.5" right, 0.5" top, and 0.5" bottom.
- Each page, except the Cover Page, Certificate of Compliance, 2D Drawing and Technical Data Sheet (TDS) shall include a page number.
- All report pages shall be ANSI A (8 1/2 x 11 inches) portrait-format.
- The Technical Design Report shall include a Table of Contents, Table of Figures, Table of Tables, Table of References and Table of Acronyms.
- The Technical Design Report shall be single-column text layout.
- The Technical Design Report shall include one Technical Data Sheet (TDS) appropriate for the team's competition entrant class. The Technical Data Sheet (TDS) must include the Team Name, School Name, and Team Number.

Table 4.3.1 Technical Design Report

	Page Count	Regular Class	Advanced Class	Micro Class
Cover Page	1	40 pts	40 pts	40 pts
Certificate of Compliance	1			
Design Report	26			
2D Drawing	1	5 pts	5 pts	5 pts
TDS: Payload Prediction	1	5 pts	-	-
TDS: Colonist Delivery Aircraft 2D Drawing	1	-	5 pts	-
TDS: Aircraft Weight Build-Up Schedule (Appendix B)	1	-	-	5 pts
Total	30	50 pts	50 pts	50 pts

4.4 2D DRAWING REQUIREMENTS

1. 2D Format and Size

The 2D drawing must be one (1) ANSI B sized page (PDF) format (11 x 17 inches).

- For teams outside North America that cannot submit an ANSI B size drawing, page format size must be the closest size available to ANSI B.

2. Markings Required

The 2D drawing must be clearly marked with:

- Team Number
- Team Name
- School Name

3. Views Required

Drawings shall include at a minimum, a standard aeronautical 3-view orthographic projection arranged as described:

- Left** side view, in lower left, with nose pointed left.
- Top** view, above and aligned with the left side view, also with nose pointed left (wing-span break-view permitted).
- Front** view aligned to side view, located in the lower right (projection view non-standard movement as noted by projection view arrows in accordance with ANSI-Y14.5M 1994).
- (Regular Class Only)** Regular Class shall include an additional view, separate from the basic aircraft, illustrating the fully loaded Cargo Bay with both Spherical Cargo and Regular Boxed Cargo. The longitudinal length of the Cargo Bay (Lcargo) must be detailed on the drawing.

4. Dimensions Required

Drawing dimensions and tolerance shall be in English units, decimal notation accordance with ANSI-Y14.5M 1994 to an appropriate level of precision to account

for construction tolerances (allowable variation from analyzed prediction to account for fabrication) (i.e. X.X = $\pm .1$ in; X.XX = $\pm .03$ in; X.XXX = $\pm .010$ in).

The minimum required dimensions/tolerances are: Aircraft length, width, and height.

5. Summary Data Required

The drawing shall contain a summary table of pertinent data to include but not limited to:

1. Wingspan
2. Empty weight
3. Battery(s) capacity
4. Motor make and model
5. Motor KV
6. Propeller manufacturer, diameter, and pitch
7. Servo manufacturer, model number and torque specification in ounce-inches for each servo used on the aircraft. Identify servo being used at each position on the aircraft.

6. Weight and Balance Information

The 2D drawing shall contain the following weight, balance and stability information:

1. A clearly marked and labeled aircraft datum
2. A weight and balance table containing pertinent aircraft equipment. Each item listed must show its location from the aircraft datum in inches (the moment arm), the force, and resultant moment. See www.saeaerodesign.com/go/downloads for additional information. The minimum list of pertinent equipment includes:
 - a. Motor
 - b. Battery(s)
 - c. Payload
 - d. Electronics
3. Aircraft mean aerodynamic cord, stability margin and Center of Gravity (CG) information listed below must be clearly shown on drawing.
 - a. Aircraft mean aerodynamic cord
 - b. Stability margin for loaded CG and empty CG
 - c. Empty CG location (flightworthy)
 - d. Fully loaded CG (flightworthy, with payload, if applicable)

4.5 TECH DATA SHEET: PAYLOAD PREDICTION (REGULAR CLASS ONLY)

Regular Class must include a total payload prediction curve as part of the technical report. The graph represents an engineering estimate of the aircraft's lift performance based on density altitude.

1. Graph of payload weight shall be linearized over the relevant range.
2. The linear equation shall be in the form of:

$$y = mX + b$$

Y = Payload weight (lbs.)
X = Density Altitude (feet)
m = Slope of the linear line
b = y-intercept.

3. Only one line and one equation may be presented on the graph. This curve may take into account predicted headwind for local conditions, rolling drag, inertia, motor and propeller performance, or any other factors that may affect take-off performance. All these factors are allowed components of the prediction curve, but only one curve will be allowed; multiple curves to account for varying headwind conditions will not be allowed.
4. The team must provide a brief explanation of how the line was generated in the body of the report. The section of the report containing this information must be noted on the payload prediction curve.
5. Graph axes shall be in English units, decimal notation.

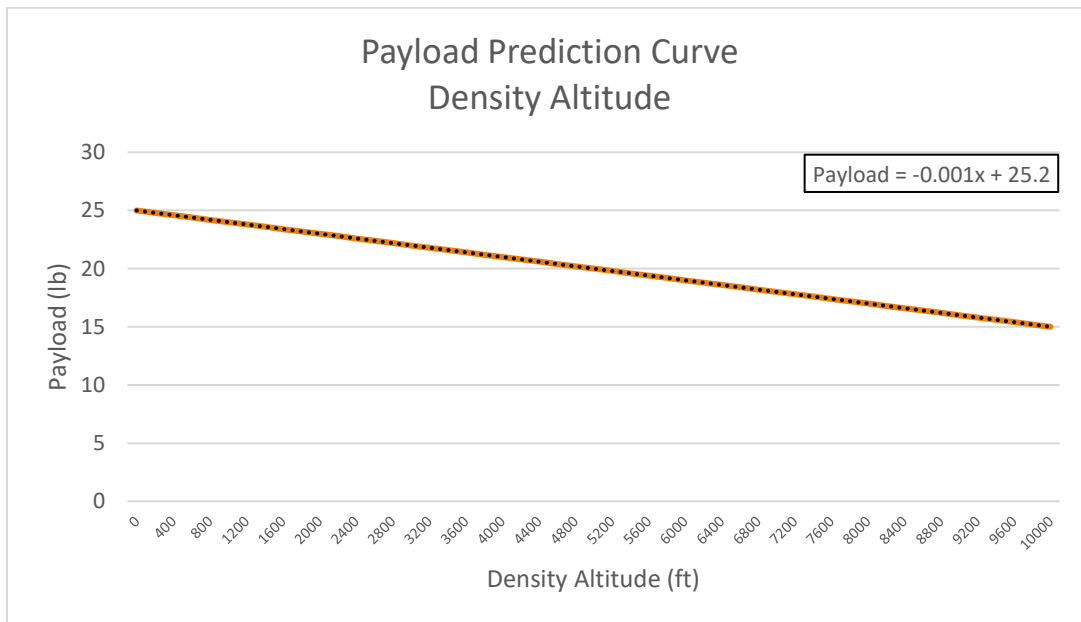


Figure 4.1: Example Regular Class Payload Prediction Curve

4.6 TECH DATA SHEET: COLONIST DELIVERY AIRCRAFT (ADVANCED CLASS ONLY)

An additional 2D drawing must be provided as an Appendix for the Colonist Delivery Aircraft (CDA). This 3-view must be ANSI B sized page (PDF) format (11 x 17 inches) and follow the same requirements as the primary aircraft 2D drawing.

1. Drawing for the Colonist Delivery Aircraft shall identify the location of the loaded CG.
2. Separate from the basic aircraft drawing, team shall provide a view illustrating the colonist cabin layout with appropriate dimensions identifying the colonist seating arrangement with total colonist capacity labeled.
3. Team shall provide a list of avionics and equipment for the Colonist Delivery Aircraft.
4. Team shall provide a glider polar (airspeed vs sink rate) for the Colonist Delivery Aircraft in fully loaded configuration, with no headwind. An explanation of the glide polar can be found on page 5-8 of the FAA Glider Handbook, Chapter 5: Glide Performance available at the following link.

https://www.faa.gov/regulations_policies/handbooks_manuals/aircraft/glider_handbook/media/gfh_ch05.pdf

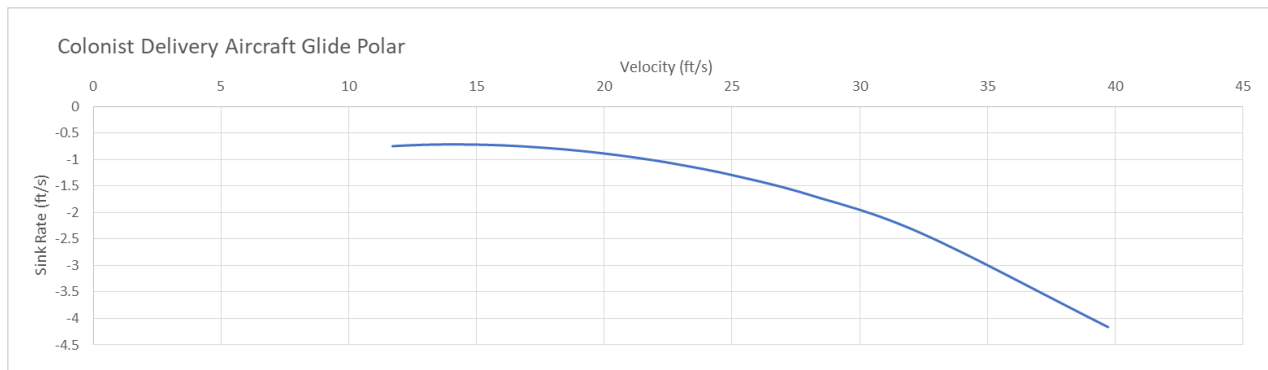


Figure 4.2: Example Advanced Class CDA Glide Polar

4.7 TECH DATA SHEET: WEIGHT BUILDUP (MICRO CLASS ONLY)

The Micro Class Weight & Balance Build-up schedule will help teams understand the importance of managing aircraft weight to achieve safety of flight at the desired payload fraction. Each team shall supply a one (1) sheet summary list of aircraft parts that contributes to the overall empty weight of the aircraft (lbs).

A template for the weight buildup can be found in Appendix B.

5 TECHNICAL PRESENTATION

Like all professionals, engineers must possess a well-developed ability to synthesize issues and communicate effectively to diverse audiences. The technical presentation portion of the aero-design competition is designed to emphasize the value of an ability to deliver clear, concise and effective oral presentations. Teams can obtain a maximum technical presentation score of fifty (50) points. The presentation score shall be comprised of scores based on the presenter's delivery technique and the judges' evaluation of technical content, empirical analysis, and visual aide.

5.1 TECHNICAL PRESENTATION REQUIREMENTS

1. Technical presentation shall last ten (10) minutes and be followed by a five (5) minute "Question and Answer" (Q&A) period.
2. Technical presentation shall be delivered in English.
3. Technical presentation shall address, but are not limited to, trade studies performed, design challenges, and manufacturing techniques.
4. Technical presentation is limited to student team members only. Non-team member pilot, Faculty Advisors, and/or parents can attend the technical presentation but are prohibited from participating in the setup, delivery, and/or the Q&A.
5. Assistance in the use of visual aids is advisable; Film clips, if used, may not exceed one-minute total duration; Film clips may not be accompanied by recorded narration.
6. All Classes shall display their entry aircraft during technical presentation.
7. Advanced class shall display one at least one (1) of each payload, including the colonist delivery aircraft.
8. During the presentation and static display setup, the teams shall provide a single sheet (8.5" x 11") marketing/promotion piece to further detail aircraft's feature, capabilities, and unique design attributes.

5.2 TECHNICAL PRESENTATION PROCESS AND PROCEDURES

Each presentation room shall have a lead judge with the responsibility to ensure compliance with competition rules and schedule. The lead judge will identify a timekeeper.

1. With agreement from the speaker, the timekeeper will give the speaker a one (1) minute warning prior to the ten (10) minute limit.
2. If the team exceeds the ten (10) minute limit, the team will be assessed a five (5) point penalty for going over the time limit.
3. The presentation shall be stopped at the eleven (11) minute mark.
4. A team shall have five (5) minutes for Q&A immediately following the presentation. Questions may be asked by any judge on the panel.
5. Any time remaining or exceeding the ten (10) minutes shall be added to or subtracted from the five (5) minute Q&A.
6. Presentation Time Breakdown:

Time (Minutes)	Description
2	Setup presentation, visual aide, and/or static display
10	Perform Technical Presentation
5	Questions & Answers
3	Pack-up presentation and static display

6 TECHNICAL INSPECTION AND AIRCRAFT DEMONSTRATIONS

Technical and Safety inspection of all aircraft will be conducted using the published Technical and Safety Inspection checklists for each class for the current year. The checklists can be found at www.saeerodesign.com/go/downloads.

Technical and Safety Inspection is the process of checking all aircraft for:

- Compliance with all general aircraft requirements.
- Compliance with all aircraft configuration requirements for their class.
- Overall safety and airworthiness.

All aircraft must pass the Technical and Safety Inspection in order to compete. **Per the Statement of Compliance, teams are required to present a fully completed Inspection checklist for their aircraft that is signed by the Faculty Advisor or Team Captain.** Teams cannot begin the inspection process without meeting this requirement. Technical and Safety inspectors at the event will confirm that the team has fully inspected their aircraft.

All required Aircraft Demonstrations will be performed at designated locations.

- **Regular Class** will demonstrate the ability to unload their aircraft within two (2) minutes per the requirements of Section 7.5. This will be demonstrated each time a team unloads the aircraft at weigh in, after each successful flight.
- **Advanced Class** will demonstrate that their aircraft has proven operational ability by providing a video showing the aircraft successfully taking off, releasing a CDA, and landing per the Section 8.1.
- **Micro Class** will demonstrate the timed assembly of their aircraft per the requirements of Section 9.6. The Timed Assembly Demonstration must be performed on Friday in the Technical Inspection and Demonstration area during the announced times. Timed Assembly Demonstrations will not be done at the flying site.

6.1 AIRCRAFT CONFORMANCE TO 2D DRAWING

During Technical Inspection, the aircraft will be inspected and measured for conformance to the 2D drawing presented in the Design Report.

1. At a minimum, aircraft length, wingspan and height dimensions will be measured and compared to the 2D drawing.
2. All teams must have a hard copy of their design report present during technical inspection.
3. Aircraft will have the actual empty CG compared to the empty CG presented in the design report 2D drawing.
4. Advanced Class must show longitudinal and lateral C.G. positions or provide a table for each payload configuration.

6.2 FAILURE TO REPORT DESIGN CHANGES

Failure to report any design changes incorporated after Design Report submission and prior to Technical Check-in will incur a one (1) point penalty for each unreported design change discovered during technical inspection.

6.3 DEVIATIONS FROM 2D DRAWING

Any deviation in construction of the aircraft from the submitted 2D drawing, after submission of the Design Report, must be reported in writing. **For Advanced and Regular Class aircraft, there is no need to report deviations in the length (L), width (W), and height (H) of the aircraft, if the following is satisfied, where dimensions are in inches:**

$$|L_{actual} - L_{drawing}| + |W_{actual} - W_{drawing}| + |H_{actual} - H_{drawing}| \leq 3 \text{ inches}$$

1. Each design change must be documented separately using the Engineering Change Request (ECR) – a physical copy of which must be brought to the Technical and Safety Inspection.
2. Only one (1) design change may be submitted per ECR form.
3. Penalty points for design changes will be assessed in accordance with the penalty guidelines in Appendix D, subject to the judges' final determination.

6.4 SAFETY AND AIRWORTHINESS OF AIRCRAFT

Technical and Safety Inspection will be also be used to assess the general safety and airworthiness aspects of each aircraft by seeking any problems that could cause an aircraft to depart controlled flight. This assessment includes, but is not limited to:

1. Unintentional wing warps
2. Control surface alignment
3. Correct control surface response to radio transmitter inputs
4. Structural and mechanical soundness

6.5 INSPECTION OF SPARE AIRCRAFT AND SPARE AIRCRAFT COMPONENTS

1. All spare aircraft and spare aircraft components (wings, fuselages and tail surfaces) must be presented for inspection.
2. Teams may submit up to two (2) complete aircraft at Technical Inspection on Friday.
3. Additional spare aircraft and parts beyond two (2) sets may be submitted for inspection during the event on Saturday and Sunday.

6.6 AIRCRAFT MUST MEET ALL INSPECTION REQUIREMENTS THROUGHOUT THE COMPETITION.

1. All aircraft must meet all Technical and Safety Inspection requirements throughout the competition.
2. Any official may request that an aircraft be re-inspected if a general, class configuration, or safety requirement problem is seen on an aircraft at any time during the event.
3. This includes any errors or omissions made by officials during inspection.

6.7 TECHNICAL AND SAFETY INSPECTION PENALTIES

No points are available to be scored as a result of the Technical and Safety Inspection: teams may only lose points as a result of errors and problems encountered during the inspection process. Any penalties assessed during Technical Inspection will be applied to the overall competition score.

7 REGULAR CLASS DESIGN REQUIREMENTS

The objective of Regular Class is to design an aircraft that can operate from short runways to carry outsized cargo as well as regular cargo. Payload will consist of large spherical storage containers, represented by Soccer Balls, and Regular Boxed Cargo, represented by payload weights, which must be carried on each flight. Accurately predicting the lifting capacity of the aircraft is an important part of the airplane design.

7.1 AIRCRAFT DIMENSION REQUIREMENT

Regular Class aircraft are limited to a maximum wingspan of 120 inches.

7.2 MATERIAL AND EQUIPMENT RESTRICTIONS FOR REGULAR CLASS

1. **Fiber-Reinforced Plastic (FRP)**

The use of Fiber-Reinforced Plastic (FRP) is prohibited on all parts of the aircraft. Fiber-Reinforced Plastic includes duct tape. Exceptions to this rule include: commercially available FRP motor mount, propeller, landing gear and control linkage components. Exploration of alternative materials is encouraged.

2. **Rubber bands**

Elastic material such as rubber bands shall not be used to retain the wing or payloads to the fuselage.

3. **Stability Assistance**

All types of gyroscopic or other stability assistance are prohibited.

7.3 AIRCRAFT SYSTEM REQUIREMENTS

1. **Electric Motor Requirements**

The aircraft shall be propelled by a single electric motor (no multiple motors). There are **no restrictions on the make or model of the electric motor**.

2. **Gear boxes, Drives, and Shafts**

Gearboxes, belt drive systems, and propeller shaft extensions are allowed if a one-to-one propeller to motor RPM is maintained. The prop(s) must rotate at motor RPM.

3. **Aircraft Propulsion System Battery**

Regular Class aircraft must be powered by a commercially available Lithium-Polymer battery pack.

1. Required: 6 cell (22.2 volt) Lithium Polymer (Li-Poly/Li-Po) battery pack. Minimum requirements for Li-Po battery: 3000 mAh, 25c
2. Homemade batteries are NOT allowed.

4. **Power Limiter**

All Regular Class aircraft must use a 2019 V2 or newer version 1000-watt power limiter from the official supplier, Neumotors.com.

1. Repair and/or modifications to the limiter are prohibited.
2. The limiter must be fully visible and easy to inspect.
3. Only battery, receiver, speed control, arming plug, and limiter are allowed within the power circuit.

4. The limiter is only available at the follow link:

<https://neumotors.cartloom.com/storefront/category/student-contests-sae-dbf>

This supplier has agreed to ship worldwide to any team.

5. Radio System Battery and Switch

If a separate battery is used for the radio system, the battery pack must have enough capacity to safely drive all the servos in the aircraft, taking into consideration the number of servos and potential current draw from those servos.

1. A battery pack with a minimum capacity of 1000 mAh must be used for the radio system.
2. The battery pack must be a LiPo or LiFE type battery.
3. Battery voltage regulators are allowed.
4. The battery pack must be controlled by a clearly visible and properly mounted on/off switch on the external surface of the aircraft, located at least 12" from the prop.

7.4 PAYLOAD REQUIREMENTS

1. Types of Cargo

Regular Class payload shall consist of two types; (1) Spherical Cargo and (2) Regular Boxed Cargo, which must both be carried internally to the aircraft. Both types of Payload must be designed for ease of access. Reference Section 7.5 for demonstration details.

2. Cargo Bay Requirements

Regular Class aircraft shall have a single fully enclosed Cargo Bay for carrying Spherical Cargo and Regular Boxed Cargo (see Section 7.4.3) with the following additional requirements:

1. The Cargo Bay shall fully enclose the Spherical Cargo and the Regular Boxed Cargo. Spherical Cargo may not be exposed to airstream at any point in flight.
2. The Cargo Bay has no restriction on size or shape.
3. Only one Cargo Bay is allowed in a Regular Class aircraft.
4. The length of the Cargo Bay (L_{cargo}) must be detailed on the drawing for Technical Inspection. The drawing must also include a schematic of the aircraft fully loaded. The length of the Cargo Bay is measured from the foremost location of any payload to the aft most location of any payload.

3. Regular Boxed Cargo Support Requirements

Regular Boxed Cargo shall consist of a support assembly and payload plates with the following additional requirements:

1. There is no required configuration for the payload plates, other than as defined by Section 2.10 and 2.11.
2. Teams must provide their own payload plates.
3. Tape, Velcro, rubber bands, container systems and friction systems alone may not be used to retain the support assembly and/or payload plates.

4. Spherical Cargo Payload Definition

The Spherical Cargo* payload must consist only of unmodified Size 5 Soccer Balls. The specifications on these Soccer Balls are:

- A circumference of not more than 28 inches and not less than 27 inches
- A weight not more than 16 ounces and not less than 14 ounces
- A pressure of 8.5 psi to 15.6 psi. While the standard says 8.5 psi, SAE Aero Design, requires a minimum of 9 psi.

Additional details can be found at:

“International football association board Rules of the Game, Law 02: the Ball”

* Teams must provide their own Spherical Cargo.

5. Spherical Cargo Carriage Requirements

Regular Class aircraft must position all Spherical Cargo in the Cargo Bay.

1. The Cargo Bay must accommodate a minimum of one (1) Spherical Cargo for each flight attempt.
2. There is no configuration requirement for the Spherical Cargo inside the Cargo Bay.

7.5 REGULAR CLASS PAYLOAD UNLOADING

Regular Class shall quickly unload both Spherical Cargo and Regular Boxed Cargo after each successful flight. This demonstration will be performed at the weigh station after the completion of each successful flight.

This is a timed activity and shall be performed by no more than two (2) members of the team within the following time constraints:

1. Two (2) minutes to unload as much Spherical Cargo and Regular Boxed Cargo as possible.
 - The demonstration will start with all Spherical Cargo and Regular Boxed Cargo loaded, secured, and the aircraft configuration unchanged from the most recent successful flight.
 - The demonstration will be considered complete at the conclusion of two (2) minutes. Any Spherical Cargo and Regular Boxed Cargo successfully unloaded from the aircraft will be weighed and recorded for scoring that flight attempt.

Any Spherical Cargo or Regular Boxed Cargo that fails to be unloaded in two (2) minutes will not be used in the scoring equation.

7.6 REGULAR CLASS SCORING

In order to participate in the flight portion of the competition, each team is required to have submitted AND received a score for their Design Report and Oral Presentation.

The team's Final Flight Score is the sum of the top three (3) flight scores the team achieves during the competition (FS_1 , FS_2 , and FS_3) and the Payload Prediction Bonus.

Scoring Equation:

$$FFS = \text{Final Flight Score} = FS_1 + FS_2 + FS_3 + PPB$$

Where:

$$FS = \text{Flight Score} = 120 * \frac{2 * S + W_{steel}}{b + L_{cargo}}$$

$$PPB = \text{Payload Prediction Bonus} = 10 - (A - P)^2$$

S = Number of Spherical Cargo Carried on a Flight

W_{steel} = Regular Boxed Cargo Weight (lbs)

b = Aircraft Wingspan (inches)

L_{cargo} = Length of Cargo Bay (inches)

A = Actual Payload = $W_{steel} + 0.9375 * S$

P = Predicted Payload

The predicted payload, P , is determined from the payload prediction curve the teams provide in the Technical Data Sheet (Section 4.5) and the density altitude measured at the event.

The Payload Prediction Bonus will be calculated for each of the top three (3) flights that are counted for score. Only the highest of these calculated bonuses will be applied to the team's final flight score.

All Payload Prediction Bonus (PPB) less than zero (0) will default to zero (0).

Penalty Points

Any penalty points assessed during the competition are now deducted from a team's overall score.

8 ADVANCED CLASS DESIGN REQUIREMENTS

The objective of Advanced Class is to design a system that can deliver colonists, habitats, and supplies to the surface of Mars. This class is focused on mission success through understanding of diverse requirements, system-level engineering, and robust execution.

8.1 VIDEO DOCUMENTATION OF PROVEN OPERATIONAL ABILITY FOR ADVANCED CLASS

All Advanced Class teams are required to bring a video documenting the proven operational ability of their Advanced Class aircraft to Technical and Safety Inspection. The hard deadline for video submission is 8AM Saturday Morning.

1. The video must show the following three activities accomplished successfully with their competition aircraft: A take-off, a successful release of a Colonist Delivery Aircraft (CDA), and a landing of the Primary Aircraft (PA) without damage to the PA. A successful release of the CDA means that the CDA is in a flyable configuration after release. No video is required of the CDA flying or landing after the release from the PA.
2. The video will be reviewed by SAE officials in the Technical Inspection area.
3. Advanced Class aircraft will not be inspected or allowed to compete without the video documentation of proven operational ability.
4. Teams must provide a device to play the video for the officials at a screen size that allows the officials to clearly see both aircrafts.
5. Videos should be no more than 1 minute in length. Edited video will be accepted if the video is of the same flight.

8.2 AIRCRAFT DIMENSION REQUIREMENT

Advanced Class aircraft are limited to a maximum wingspan of **132 inches**.

8.3 AIRCRAFT SYSTEM REQUIREMENTS

1. Electric Motor Requirements

The Primary Aircraft shall be propelled by one or more electric motors. There are no restrictions on the make or model of the electric motor.

2. Gear boxes, Drives, and Shafts

Gearboxes, belt drive systems, and propeller shaft extensions are allowed.

3. Aircraft Propulsion System Battery

Advanced Class Primary Aircraft shall be powered by a single commercially available Lithium-Polymer battery pack.

1. Required: 6 cell (22.2 volt) Lithium Polymer (Li-Poly/Li-Po) battery pack. Minimum requirements for Li-Po battery: 3000 mAh, 25c
2. Homemade batteries are NOT allowed.

4. Power Limiter

All Advanced Class Primary Aircraft shall use a single 2018 or newer version 750-watt power limiter from the official supplier, Neumotors.com.

1. Repair and/or modifications to the limiter are prohibited.
2. If multiple motors are used, all throttle signals and power must pass through a single limiter.
3. The limiter must be fully visible and easy to inspect.
4. Only battery, receiver, speed control, arming plug, and limiter are allowed within the power circuit.
5. The limiter is only available at the follow link:

<https://neumotors.cartloom.com/storefront/category/student-contests-sae-dbf>

This supplier has agreed to ship worldwide to any team.

8.4 RADIO SYSTEM BATTERY

The radio system battery pack must have enough capacity to safely drive all the servos in the Primary Aircraft, taking into consideration the number of servos and potential current draw from those servos. If the radio system battery also supplies DAS or other power needs, the radio system battery must be large enough for these power requirements as well.

1. A battery pack with a minimum capacity of 1000 mAh must be used for the radio system.
2. The battery pack must be a LiPo or LiFE type battery.
3. Battery voltage regulators are allowed.
4. The battery pack must be controlled by a clearly visible and properly mounted on/off switch on the external surface of the Primary Aircraft, located at least 12" from the prop.

8.5 RUBBER BANDS

Rubber bands shall not be used to retain the wing to the fuselage.

8.6 COLONIST DELIVERY REQUIREMENTS

Teams are responsible for delivering Colonists safely to the surface of the planet through up to three (3) unpowered (no on-board propulsion) autonomously guided aircraft.

1. Colonists

Colonists must consist only of unmodified table tennis (ping-pong) balls which meet or exceed the minimum size and weight specifications as specified by the International Table Tennis Federation Technical Leaflet T3: The Ball. Accordingly, the minimum ping-pong ball weight is 0.095 ounces and minimum ball diameter allowed is 1.57 inches. Teams must provide their own Colonists.

2. Colonist Delivery Aircraft (CDA)

The following requirements apply to the colonist delivery aircraft:

1. Total weight of each CDA must be less than 9.0 oz.
2. Center of gravity must be clearly marked.
3. If more than one CDA is used, each must be uniquely colored, visible from the ground. Each CDA's transmitter must be marked with the matching color.
4. All Colonists must be carried internally in a single colonist cabin.
5. The colonist cabin must position all Colonists to be tangent to the same side of a single geometric plane.
6. Colonists must be in a countable configuration. A countable configuration is defined as when Colonists are clearly visible and can be easily touch-counted.
7. If the number of Colonists is changed from the 2D drawing, there is a one (1) point penalty per Colonist.
8. Three (3) Teladrop® resettable impact shock sensors rated at **50G** must be rigidly mounted to the airframe within 1 inch of the CG: 1 Sensor in z-direction, 1 sensor in the y-direction (span), 1 sensor in x-direction (forward/aft). These sensors must be provided by each team and are available at the following links:

<https://www.telatemp.com/p/415/50g-teladrop-drop-n-tell-damage-indicator>

<https://www.shippinglabels.com/teladrop-drop-n-tell-resettable-shock-indicator/sku-l-drop-n-tell-r>

9. CDA(s) will be considered part of the mothership prior to the intentional release and separation towards the target landing zone. Section 3.5 will be observed if the CDA(s) lose parts while attached to the mothership. Structural components will result in a disqualification of the flight attempt. Non-structural components will result in a 25% penalty.

8.7 SUPPLY PAYLOAD REQUIREMENTS

There is no limit to the number of Supply Payloads that can be carried by each Primary Aircraft. The following requirements apply to each Supply Payload:

1. Supply Payloads may be mounted internally and/or externally to the airframe.
2. The CG of each mounted Supply Payload shall not exceed ten (10) inches longitudinally from the Primary Aircraft's EW CG. This shall be measured during Technical Inspection.
3. All Payloads that are connected (e.g. become tangled during drop), in any way, shall not be counted for score.
4. All Supply Payloads that achieve a scoring drop on the target, or a successful drop from above 100 feet that misses the target, will be inspected and measured at the weigh station after that drop. This is the only time Supply Payloads will be inspected. If any Supply Payload is not in compliance with all related rules, score for that Supply Payload will be zero (0) for that flight attempt.

1. Habitat Module

A Habitat Module must consist of an unmodified Nerf Sports Vortex Aero Howler.

1. Each Habitat Module shall be marked with the team number measuring 2" in height.
2. If a team elects to make two separate habitat drops during a flight attempt, the habitats for each separate drop shall be a different color.

2. Water Bottle

Water delivery must consist of unopened plastic bottles of liquid water. Ice is not permitted.

1. Allowable water bottles are any commercially available bottle with size of 16.9 FL Oz (500 ml) or 33.8 FL Oz (1 liter).
2. Each water bottle shall have one (1) flexible streamer attached to facilitate payload location and recovery.
3. Each streamer shall be at least 48 inches long and 2 inches wide.
4. Water bottles shall be marked with the team number in two (2) places: on the container, and one end of the streamer with numbers 2-inches in height.
5. Streamers must always remain attached to each water bottle. Attached is defined as being able to support the weight of each water bottle.
6. If a team elects to make two independent drops during a flight attempt, the water bottles for each independent drop shall have streamers of a different color.
7. Streamers shall be in a stored configuration prior to the drop and deploy before each water bottle strikes the ground.
8. The original label must be left on the bottle for inspection. If the label is damaged or comes off during a round, the team will not be penalized.

8.8 STATIC PAYLOAD REQUIREMENTS

1. Static payload shall be in its own payload bay(s).
2. Static payload bay(s) shall be fully enclosed, completely closed off, and physically separated from all releasable payload.
3. Static payload bay(s) shall have no restriction on size or shape.
4. Teams must be able to unload Static Payload at the weigh station after the flight in five (5) minutes or less.

8.9 GYROSCOPIC AND OTHER STABILITY AUGMENTATION

Gyroscopic assist or other forms of stability augmentation are allowed in Advanced Class.

8.10 AUTONOMOUS FLIGHT

Autonomous flight systems that cause the Primary Aircraft to navigate without direct pilot control input are prohibited.

Autonomous flight for the CDA(s) is required, subject to the following rules:

1. At least two degrees of freedom must be controlled by the autonomous system.
2. Teams must have a manual override for control over the CDA through secondary transmitters. This may be a switch on that transmitter to select between autonomous and manual flight modes.
3. Manual override will be used at the discretion of the Air Boss if there are safety concerns. Any use of the manual override disqualifies the CDA from successfully scoring points.
4. When switching to manual flight override, the CDA should immediately actuate full up on pitch control. No other manual control or flying will be allowed.

8.11 DATA ACQUISITION SYSTEM (DAS)

Advanced Class Primary Aircraft must have a Data Acquisition System (DAS) that shall record altitude and be used by the payload specialist to guide the primary pilot towards the target zone.

1. Using a ground receiver station, the team must display the real-time altitude of the aircraft to the Payload Specialist and the flight judge.
2. Team must automatically record, and immediately display the altitude (ft) at the moment when a CDA or supplies are released from the Primary Aircraft. **A half inch (0.5") indicator must state which type of payload (CDA, water, or habitat) was released and provide in half inch (0.5") font, a single altitude associated with each released payload.** The indicator must remain visible for the remaining duration of the flight.
3. The DAS recording must be performed on the ground station and must support play back for review on demand.
4. Altitude must be measured in feet with display precision of at least one (1) ft. and an accuracy error of less than ten (10) ft.
5. DAS system must use a discrete and removable Red arming plug to apply power to the DAS system. The DAS arming plug must be located on top of the Primary Aircraft at least 12 inches from the propeller. One Red arming plug can be used for both DAS and FPV. If the DAS and Aircraft Propulsion System Arming plugs are different, both must be removed upon landing to minimize interference with other teams.
6. DAS equipment may also have a reset switch, if desired. If a manual reset switch is used, it must be located externally at least 12 inches away from the propeller. A wireless DAS reset system is allowed.
7. DAS systems shall not use the same 2.4 GHz channel as the flight control system, unless the telemetry being used is part of the radio control system being used. A DAS built into the radio control system must meet all DAS rules requirements.

8.12 FIRST PERSON VIEW SYSTEM (FPV)

FPV is no longer required for Advanced Class. For teams that wish to use an FPV system for operational reasons, the following conditions apply:

1. Teams will be required to sign up for one of 12 discrete commonly used FPV frequencies. The frequency list will be provided by SAE Aero Design.

2. There will be a frequency sign-up process communicated to teams via the event newsletters.
3. If more than 12 Advanced Class teams choose to use an FPV system, some team's frequencies may have more than 1 team using them. Frequency control procedures will be in place at the event to prevent conflicts.
4. The primary pilot must fly visually only (no FPV goggles or ground station reference).
5. FPV systems CANNOT use the same frequency as the flight control system. Use of 2.4 GHz for FPV video is prohibited.
6. The FPV system must use a discrete and removable Red arming plug to apply power to the FPV system. The Red arming plug must be located on top of the aircraft at least 12 inches from the propeller. One Red arming plug can be used for both DAS and FPV.

8.13 DAS FAILURES

Any DAS failure during the flight attempt is considered a missed flight attempt and receives zero (0) points.

Example: A team has flown four (4) rounds successfully and on the 5th round the Primary Aircraft takes-off successfully, makes a successful drop, but the DAS altitude reading malfunctions. The flight attempt will NOT be considered a qualified flight and the team will receive zero (0) credit for colonist, supplies or static payload for round 5.

8.14 PAYLOAD SPECIALIST

Advanced Class Primary Aircraft must be able to release payloads using a system commanded by the Payload Specialist.

1. The Payload Specialist must be a team member and must verbally direct the pilot, with altitude and direction cues, to the drop zone using only the DAS. Only one person can serve as Payload Specialist and communicate with the Pilot. The Pilot may not communicate with the Payload Specialist except to request a repeat of an instruction. The Payload Specialist should not count on having a line-of-sight view to the aircraft.
2. Teams are allowed two (2) drop attempts on the target.
3. The primary pilot cannot have access to or activate any payload release, and the release cannot be connected to the primary pilot's R/C transmitter in any way.
4. The payload release must be manually activated by the Payload Specialist or by an automatic release system that is part of the Primary Aircraft electronics.
5. If an automatic payload release system is used, it must have a manual override controlled by the Payload Specialist.
6. Teams may activate the payload release system using a second 2.4 GHz radio system or some other method based on their DAS or telemetry system.

8.15 COLONIST DELIVERY AIRCRAFT RELEASE PROCEDURES

1. Release of the CDA(s) must be at least 200 feet away the center of the drop zone, measured parallel to the runway. An SAE official with flags will be stationed up and down wind of the drop zone to indicate when the Primary Aircraft is outside this distance.
2. Teams must release CDA(s) at an altitude of above 50 ft but no higher than 100 ft.
3. Teams may release the CDA(s) at any point in their flight pattern, as long as they adhere to 8.15.1 and 8.15.2. See Figure 8.1.
4. If any of the mounted shock sensors on the CDA show that they have been tripped upon landing, no points will be awarded for the Colonists on that CDA.

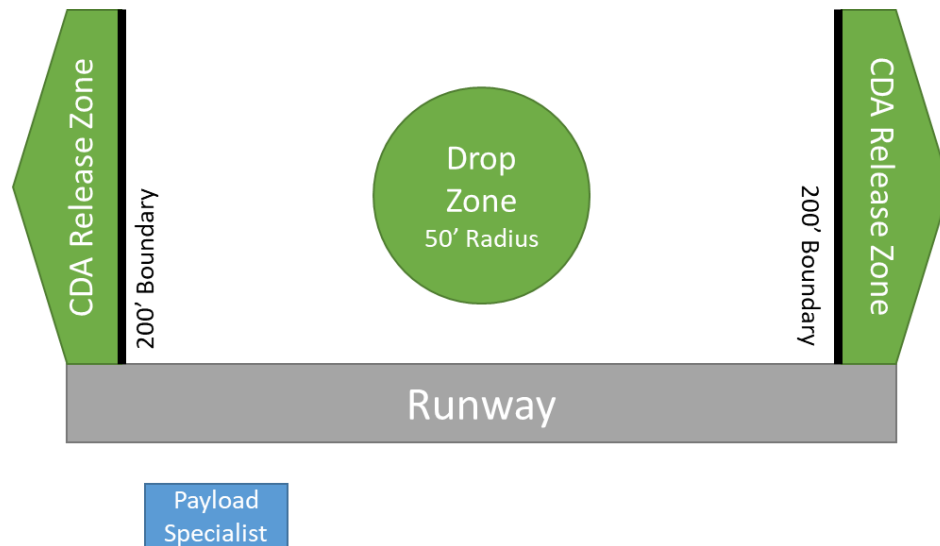


Figure 8.1 – CDA Release Zone

8.16 SUPPLY DROP PROCEDURES

1. Teams can drop the Supply Payloads in the upwind or downwind directions or both.
2. Dropping direction shall be declared to the Air Boss prior to take-off and adhered to during flight operations. Stated direction will be recorded on the flight log.
3. Each approach to the target in the direction declared to the Air Boss is considered a drop attempt. Teams may drop as many of each payload type as they wish during each attempt.
4. Teams must drop Supply Payloads at an altitude of above 100ft.
5. To receive credit for a flight, the team shall successfully release at least one (1) type of payload (CDA, water, or habitat) at its required altitude.

A successful payload release is defined as the intentional targeted drop of at least one payload (CDA, water, or habitat), while conforming to each payload's rules as found in Sections 8.15 and 8.16. The payload does not have to land in the target zone but must meet all requirements after the drop. Altitude at release must be successfully recorded on the DAS and displayed on the ground station.

Example 1:

A team states they will drop in one direction (upwind or downwind). They will get two (2) attempts to position their Primary Aircraft in the proper direction. Each approach to the drop zone in the declared direction will be counted as a drop attempt.

Example 2:

A team states they will drop in both directions; upwind and downwind. Any approach to the drop zone is considered an attempt.

8.17 ADVANCE CLASS SCORING

In order to participate in the flight portion of the competition, each team is required to have submitted AND received a score for both Design Report and Oral Presentation.

The final flight score is based on the total quantity of Colonists and Days of Habitability, derived from the number habitats and amount of water a team successfully delivers across the entire event. Delivery consists of the Colonist Delivery Aircraft or Supply Payloads coming to rest undamaged within a defined 50ft radius circle.

Teams must deliver sufficient habitats and water to support the growing population of their colony. For example, a team that delivers 4 habitats but no water will achieve a Days of Habitability of 0. Additional points are awarded for static payload.

Scoring Equation:

$$FFS = \text{Final Flight Score} = \frac{N_C \times D}{15N} + \frac{2S_p}{N}$$

Where:

$$D = \text{Days of Habitability} = 25 \left(2^{1 - \text{Maximum} \left(\frac{N_C}{8N_H}, \frac{N_C}{N_W} \right)} \right)$$

N = Total Number of Flight Rounds Offered During the Competition

N_C = Total Number Colonists Delivered During the Competition

N_H = Total Number of Habitats Delivered During the Competition

N_W = Total Amount of Water (in fl oz) Delivered During the Competition

S_p = Total Static Payload (lbs) Delivered During the Competition

Note: If 5 flight rounds are offered, but a team only flies in 3 rounds, N is still 5.

Penalty Points

Any penalty points assessed during the competition are now deducted from a team's overall score.

9 MICRO CLASS DESIGN REQUIREMENTS

The objective of Micro Class is to design light-weight micro UAV style aircraft that can be quickly deployed from a small package and able to carry a large, unwieldy low-density payload. The scores are based on both the maximum performance of the aircraft and on overall performance throughout the event. The Micro Class assembly demo is a mandatory timed event. Payload fraction is still a core element of the class.

9.1 AIRCRAFT SYSTEMS REQUIREMENTS

1. Propulsion Requirements

Micro Class aircraft are restricted to electric motor propulsion only.

2. Propeller and Gearbox

Gearboxes on a Micro Class aircraft where the propeller RPM differs from the motor RPM are allowed. Multiple motors, multiple propellers, propeller shrouds, and ducted fans are allowed in Micro Class.

3. Aircraft Propulsion System Battery

Micro Class aircraft must use Lithium Polymer batteries. The maximum size propulsion system battery allowed for Micro Class is a 3 cell 2200mAh lithium polymer battery. Batteries having fewer cells and lower capacity are permitted.

4. Gyroscopic Assist Allowed

Gyroscopic assist and other forms of stability augmentation are allowed in Micro Class.

5. Aircraft Empty Weight Definition

An empty aircraft has completed a successful flight and has the payload removed for weigh-in. All aircraft parts that are not payload, as defined in Section 9.2, contribute to the empty aircraft weight, including, but not limited to: airframe, receiver, electronics, batteries, hardware, brackets, straps and other associated features.

9.2 PAYLOAD REQUIREMENTS

1. Micro Class aircraft shall use 2-inch diameter, Schedule 40, White Polyvinyl Chloride (PVC) pipe in accordance with ASTM D1785 as payload weights:
 1. Outer diameter: 2.375 inches
 2. Nominal Inner diameter: 2.000 inches
 3. Minimum Wall Thickness: 0.154 inches
 4. Inner diameter (Ref): 2.067 inches MAX
 5. Weight (Ref): 0.680 lbm. /ft.
 6. Color: White
2. When free from the aircraft, the inner diameter of the payload pipes shall be free from obstruction.
3. Except being cut to length or having holes drilled in the sidewall of the pipe, the PVC pipes shall be unmodified.
4. Payload support structure for the PVC pipes is NOT included in the payload weight for scoring purposes.

Notes:

- There is no requirement for internal carriage of payload within the aircraft.
- There is no required carriage configuration for the payload.

9.3 MICRO CLASS AIRCRAFT LAUNCH

1. Hand launched

The Micro Class aircraft must be launched by hand using an overhand motion.

1. The pilot and one (1) member of the team can be on the runway to launch the aircraft.
2. The aircraft can only be launched by one (1) team member and must not be the pilot.
3. The pilot must be outside the pre-marked launch zone during the launch.
4. There is no limit on number of steps taken during the overhand launch but the person launching must remain in the launch zone before and after the overhand launch. Stepping outside the launch zone will invalidate the flight attempt.
5. The team member performing the launch action must always have at least one foot on the ground.

2. Overhand launching violations

The following actions are not permitted and will invalidate the flight attempt and score for the round:

4. Overhand launching the aircraft from any other part of the aircraft other than the fuselage or payload attached to the fuselage
5. Running with the aircraft during launch
6. Pilot or non-team member launching the aircraft

9.4 MICRO CLASS AIRCRAFT HAND-LAUNCH SAFETY REQUIREMENTS

Safety gear must be used by the designated team member performing the aircraft launch and any team member assisting with preparing the aircraft.

Safety gear will consist of:

- Safety glasses
- Hard hat

9.5 AIRCRAFT SYSTEM CONTAINER

1. Aircraft System Container Requirements

Micro Class aircraft will fit in an aircraft system container with size limitations. Compliance with the following requirements will be confirmed during Technical Inspection.

1. The aircraft system container shall be a cardboard box with absolute maximum outside dimensions of 12.125 inches X 3.625 inches X 13.875 inches maximum. Minimum wall thickness of container is 0.125 inches.

2. The fully packed aircraft system container must not weigh more than 10 pounds (lbs.).
3. The aircraft system container must be labeled with the school name, school address, team name, and team number.

2. Aircraft System Packaging General Requirements

The aircraft container must contain the following:

1. The following items must be packaged within the aircraft system container:
 - The airframe
 - Propulsion system battery
 - All (maximum) payload to be carried. Teams may not carry any payload that was not stored in the aircraft container.
 - All tools needed for the Assembly Demonstration, except a box-cutter or knife to cut packing tape
2. Any additional components not listed herein required for flight. The propulsion system battery must not be pre-installed in the aircraft.
3. The red arming plug must not be pre-installed in the aircraft.
4. The transmitter and any spare parts are not required to be in the aircraft system container.
5. The payload shall not be pre-installed to the aircraft.

9.6 TIMED AIRCRAFT ASSEMBLY DEMONSTRATION

1. Performance

The Assembly Demonstration must be performed on Friday in the Technical Inspection and Demonstration area during the announced times. Timed Assembly Demonstrations will not be done at the flying site.

The Assembly Demonstration shall be accomplished under the following constraints:

1. The assembly of the aircraft shall be accomplished within three (3) minutes. If 3 minutes is exceeded, then the demonstration will be halted.
2. The Assembly Demonstration score (AD) will be calculated per Section 9.8.
3. The aircraft shall be completely packaged in the aircraft system container. The box will be taped closed with packing tape.
4. The Assembly Demonstration will consist of aircraft removal from the box and installation of the propulsion system battery at a minimum.
5. The use of any type of time-dependent curing adhesives such as glue, super-glue or epoxies are prohibited in the Assembly Demonstration.
6. Only the tools contained within the aircraft system container at the beginning of the Assembly Demonstration are permitted for use during the assembly process.
7. All payload carried in the aircraft system container will be installed on the aircraft before the Assembly Demonstration is considered complete.

2. Process for Assembly Demonstration:

1. Two (2) team members tasked with assembling the aircraft will be located at a table in the Technical Inspection and Demonstration area. At this time, the fully packaged, non-energized aircraft with flight battery uninstalled, will be in the container with the aircraft system container taped closed. Failure to have the Red Arming Plug removed or the battery uninstalled will result in an assembly demonstration time of three (3) minutes.
2. An official will give the “GO” command to begin the Assembly Demonstration. Assembly will start with the team opening the aircraft system container. Officials will record the elapsed assembly time. Teams may use a knife to cut open the aircraft system container as described in rule 9.5.2.1.
3. The prop will be installed for the demo but shall not be fully tightened to the motor for safety reasons.
4. When the aircraft is fully assembled, with all (maximum) payload from the container installed, with the flight battery installed, the team will give the “DONE” command to signal the officials to stop timing.
5. After the “DONE” command is given by the assembling team, no further assembly may occur.
6. The Assembly Demonstration is considered complete when all tasks required for flight have been performed except for:
 - Installing the Red Arming Plug
 - Performing preflight controls check
7. The officials will inspect the aircraft to confirm aircraft flight ready status and record elapsed time.
8. The aircraft will then be powered up for a flight control and motor function check. If the aircraft checks out as fully ready for flight, the Assembly Demonstration time is awarded.
9. Disassembling the aircraft during the pre-flight and motor checks will invalidate any assembly time of less than three (3) minutes.

9.7 MISSION REQUIREMENTS

1. Time Limit for Aircraft Launch

Micro Class aircraft should be assembled prior to stepping onto the runway.

1. Each team will have 60 seconds to complete preflight checks, energize the propulsion system, and check the controls and hand-launch the aircraft.
2. Only one (1) launch attempt is permitted per round.

2. Aircraft Take-off and Circuit

Micro Class Take-off is defined as the point at which the aircraft departs the hand of the person throwing the aircraft. Once take-off occurs, Micro Class aircraft are required to:

1. Remain airborne and fly past the designated turn point before turning approximately 180 degrees in heading. The distance required before initial turn will be field specific and discussed at the Friday briefing and pilot's meetings. Micro Class aircraft are expected to be able to complete a 400-foot leg of a flight circuit.
2. Fly past a second designated turn point, turning 180 degrees in heading.
3. Land in the designated landing zone. Micro Class aircraft should be prepared to land on either a paved landing zone or an unpaved landing zone.
4. Take-off direction will be determined by the Air Boss, and normally selected to face into the wind.

9.8 MICRO CLASS FLIGHT SCORING

In order to participate in the flight portion of the competition, each team is required to have submitted AND received a score for both Design Report and Oral Presentation.

Final Flight Score will be calculated in two parts: the average of all flight round scores and the maximum scored for any one flight round.

Scoring Equation:

$$Final\ Flight\ Score = FSS = 20 * \left[0.5 * \left(\frac{1}{N} \sum_{1}^N FS_n \right) + 0.5 * MAX(FS_n) \right] + AD$$

Where:

$$FS_n = Flight\ Score_n = \frac{W_{payload}}{\sqrt{W_{empty}}}$$

The weight used for the scoring equation will be in pounds (lbs).

The Micro Class Assembly Demo score, either positive or negative, will be added to or deducted from the team's overall score, per Section 3.16.

Assembly Demonstration

$$AD = Assembly\ Demonstration = 5 * \left(2 - \frac{t}{60} \right)^3$$

MAX = Team's maximum single flight round score

t = time recorded in seconds

N = total number of flight rounds offered during the competition

Note: If 5 flight rounds are offered, but a team only flies in 3 rounds, N is still 5.

Penalty Points:

Any penalty points assessed during the competition will be deducted from the team's overall score.

APPENDIX A

STATEMENT OF COMPLIANCE

Certification of Qualification

Team Name _____ Team Number _____
School _____
Faculty Advisor _____
Faculty Advisor's
Email _____

Statement of Compliance

As faculty Adviser:

_____ (Initial) I certify that the registered team members are enrolled in collegiate courses.

_____ (Initial) I certify that this team has designed and constructed the radio-controlled aircraft in the past nine (9) months with the intention to use this aircraft in the **2020** SAE Aero Design competition, without direct assistance from professional engineers, R/C model experts, and/or related professionals.

_____ (Initial) I certify that this year's Design Report has original content written by members of this year's team.

_____ (Initial) I certify that all reused content have been properly referenced and is in compliance with the University's plagiarism and reuse policies.

_____ (Initial) I certify that the team has used the Aero Design inspection checklist to inspect their aircraft before arrival at Technical Inspection and that the team will present this completed checklist, signed by the Faculty Advisor or Team Captain, to the inspectors before Technical Inspection begins.

Signature of Faculty Advisor

Date

Signature of Team Captain

Date

Note: A copy of this statement needs to be included in your Design Report as page 2 (Reference Section 4.3)

APPENDIX B

Micro Class Tech–Data Sheet: Weight Buildup

Team Name:

School Name:

Team Number:

	Component	Weight (lb.)
1		
2		
3		
4		
5		
6		
7		
8		
9		
10		
11		
12		
13		
14		
15		
16		
17		
18		
19		
20		
21		
22		
23		
24		
25		
26		
27		
28		
29		
30		
31		
32		
33		
34		
Total		

APPENDIX C

Engineering Change Request (ECR)

Team Number:				
School Name:				
Team Name:				
Discovery Method	<input type="checkbox"/> Tech Inspection <input type="checkbox"/> Safety Inspection <input type="checkbox"/> Test Flight <input type="checkbox"/> Design Analysis	System Affected	<input type="checkbox"/> Wing (area +/-) <input type="checkbox"/> Fuselage (area +/-) <input type="checkbox"/> Horiz. Stabilizer (area +/-) <input type="checkbox"/> Vertical Tail (area +/-) <input type="checkbox"/> Engine Mount assembly	<input type="checkbox"/> Mechanical <input type="checkbox"/> Landing System <input type="checkbox"/> Structural <input type="checkbox"/> Electronics (avionics) <input type="checkbox"/> Cargo Bay Assembly
Surface Area	AREA ADDED: _____ AREA REDUCED: _____ <i>If surface area was impacted by the modification, specify total area added or reduced. Show calculations:</i>			
Dimensions Modified	<u>Original Dimension:</u> _____ Modified Dimension: _____			
Describe the Modification				
Reason for Modification				
Other Considerations				
*** OFFICIAL USE ONLY ***				
ECR #				

APPENDIX D

Penalty Chart Guidelines

These charts provide guidelines to possible assessment of penalty points for different design changes. Final assessment of penalty points is subject to the judges' determination.

Table D1: Penalties guidelines for for wing surface changes

Dimension	Add	Remove
Span	2pts per inch	1pt per inch
Chord	10pts per inch	5 pts per inch

For Advanced and Regular Class aircraft, there is no penalty for deviations in the length (L), width (W), and height (H) of the aircraft, if the following is satisfied, where dimensions are in inches:

$$|L_{actual} - L_{drawing}| + |W_{actual} - W_{drawing}| + |H_{actual} - H_{drawing}| \leq 3 \text{ inches}$$

Table D2: Penalty guidelines by category and size of change

Type	Small	Medium	Large
Structural	2pts	4pts	6pts
Mechanical	2pts	4pts	6pts
Electronics	1pts	2pts	3pts
Miscellaneous	1pts	3pts	5pts

APPENDIX E

APPEALS

Team Name	
Team Captain	
Collateral Points	<p><i>All appeals will require the team to post twenty-five (25) points as collateral. If the appeal is successful and the action is reversed, the team will not forfeit the twenty-five (25) collateral points. If the appeal is overruled, the team will forfeit the twenty-five (25) collateral points</i></p> <p>Collateral Points: <input type="text" value="25"/></p> <p>Sign if Agree: _____</p>
Reason for this Appeal	
Rule Reference	<p><i>List the section(s) in the official rule that is (are) in conflict with the action(s) taken by competition official</i></p> <p>Section: _____ Section: _____</p> <p>Section: _____ Section: _____</p>
Desire outcome	

APPENDIX F

SAE Technical Standards

The SAE Technical Standards Board (TSB) has made the following SAE Technical Standards available on line, **at no cost**, for use by Collegiate Design teams. Standards are important in all areas of engineering and we urge you to review these documents and to become familiar with their contents and use.

The technical documents listed below include both (1) standards that are identified in the rules and (2) standards that the TSB and the various Rules Committees believe are valuable references or which may be mentioned in future rule sets.

All Collegiate Design Series teams registered for competitions in North America have access to all the standards listed below - including standards not specific to your competition.

SAE Technical Standards included in the CDS Rules

Baja SAE

J586 - Stop Lamps for Use on Motor Vehicles Less Than 2032 mm in Overall Width

J759 - Lighting Identification Code

J994 - Alarm - Backup – Electric Laboratory Tests

J1741 - Discriminating Back-Up Alarm Standard

Clean Snowmobile Challenge

J192 - Maximum Exterior Sound Level for Snowmobiles

J1161 - Sound Measurement – Off-Road Self-Propelled Work Machines Operator-Work Cycle

Formula Hybrid

J1318 - Gaseous Discharge Warning Lamp for Authorized Emergency, Maintenance and Service Vehicles

J1673 - High Voltage Automotive Wiring Assembly Design

Formula SAE

SAE 4130 steel is referenced but no specific standard is identified

SAE Grade 5 bolts are required but no specific standard is identified

Supermileage

J586 - Stop Lamps for Use on Motor Vehicles Less Than 2032 mm in Overall Width

SAE Technical Standards for Supplemental Use

Standards Relevant to Baja SAE

J98 – Personal Protection for General Purpose Industrial Machines – Standard

J183 – Engine Oil Performance and Engine Service Classification - Standard

J306 – Automotive Gear Lubricant Viscosity Classification - Standard

J429 – Mechanical and Material Requirements for Externally Threaded Fasteners – Standard

J512 – Automotive Tube Fittings - Standard

J517 – Hydraulic Hose - Standard

J1166 – Sound Measurement – Off-Road Self-Propelled Work Machines Operator-Work Cycle

J1194 – Rollover Protective Structures (ROPS) for Wheeled Agricultural Tractors

J1362 – Graphical Symbols for Operator Controls and Displays on Off-Road Self-Propelled Work Machines - Standard

J1614 – Wiring Distribution Systems for Construction, Agricultural and Off-Road Work Machines

J1703 - Motor Vehicle Brake Fluid - Standard

J2030 – Heavy Duty Electrical Connector Performance Standard

J2402 – Road Vehicles – Symbols for Controls, Indicators and Tell-Tales – Standard

Standards Relevant to Clean Snowmobile Challenge

J44 – Service Brake System Performance Requirements – Snowmobiles - Recommended Practice

J45 – Brake System Test Procedure – Snowmobiles – Recommended Practice

J68 – Tests for Snowmobile Switching Devices and Components - Recommended Practice

J89 – Dynamic Cushioning Performance Criteria for Snowmobile Seats - Recommended Practice

J92 – Snowmobile Throttle Control Systems – Recommended Practice

J192 – Maximum Exterior Sound Level for Snowmobiles - Recommended Practice

J288 – Snowmobile Fuel Tanks - Recommended Practice

J1161 – Operational Sound Level Measurement Procedure for Snowmobiles - Recommended Practice

J1222 – Speed Control Assurance for Snowmobiles - Recommended Practice

J1279 – Snowmobile Drive Mechanisms - Recommended Practice

J1282 – Snowmobile Brake Control Systems - Recommended Practice

J2567 – Measurement of Exhaust Sound Levels of Stationary Snowmobiles - Recommended Practice

Standards Relevant to Formula SAE

J183 – Engine Oil Performance and Engine Service Classification - Standard

J306 – Automotive Gear Lubricant Viscosity Classification - Standard

J429 – Mechanical and Material Requirements for Externally Threaded Fasteners – Standard

J452 - General Information – Chemical Compositions, Mechanical and Physical Properties of SAE Aluminum Casting Alloys – Information Report

J512 – Automotive Tube Fittings - Standard

J517 – Hydraulic Hose - Standard

J637 – Automotive V-Belt Drives – Recommended Practice

J829 – Fuel Tank Filler Cap and Cap Retainer

J1153 - Hydraulic Cylinders for Motor Vehicle Brakes – Test Procedure

J1154 – Hydraulic Master Cylinders for Motor Vehicle Brakes - Performance Requirements - Standard

J1703 - Motor Vehicle Brake Fluid - Standard

J2045 – Performance Requirements for Fuel System Tubing Assemblies - Standard

J2053 – Brake Master Cylinder Plastic Reservoir Assembly for Road Vehicles – Standard

Standard Relevant to Formula Hybrid

J1772 – SAE Electric Vehicle and Plug in Hybrid Conductive Charge Coupler

Standard Relevant to all CDS Competitions

J1739 – Potential Failure Mode and Effects Analysis in Design (Design FMEA) Potential Failure Mode and Effects Analysis in Manufacturing and Assembly Processes (Process FMEA) and Potential Failure Mode and Effects Analysis for Machinery (Machinery FMEA)